

# Advances In High Temperature (Viscoelastoplastic) Material Modeling for Thermal Structural Analysis



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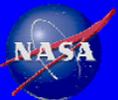
# ***OUTLINE***

- **Background/Philosophy**

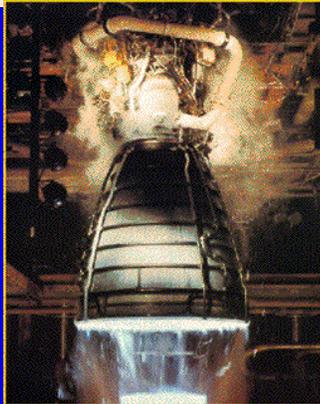
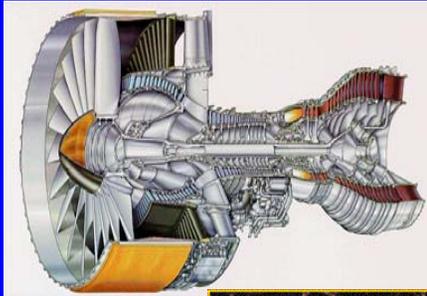
- Elevated Material Behavior
- Impact on Analysis
- Multiscale Framework/Vision

- **Recent Advances**

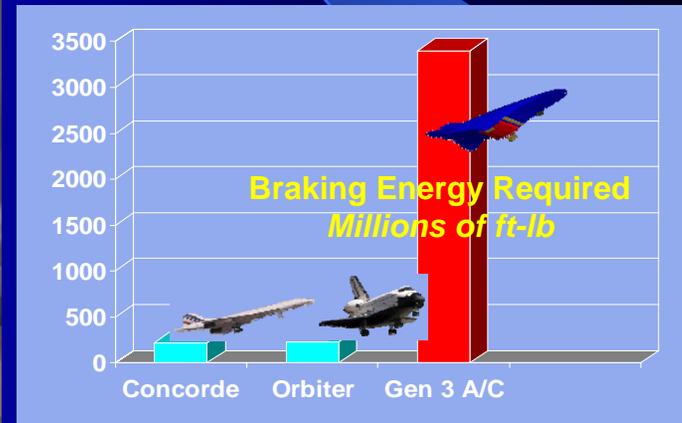
- Theoretical Modeling/Testing
- Numerical Integration
- Material Characterization



# Typical High Temperature Applications Demand High Performance Materials



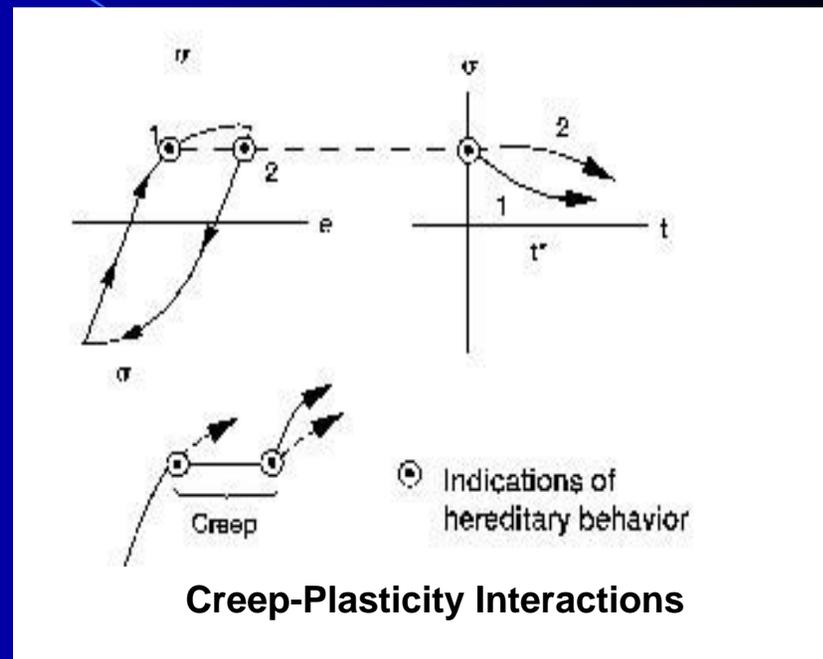
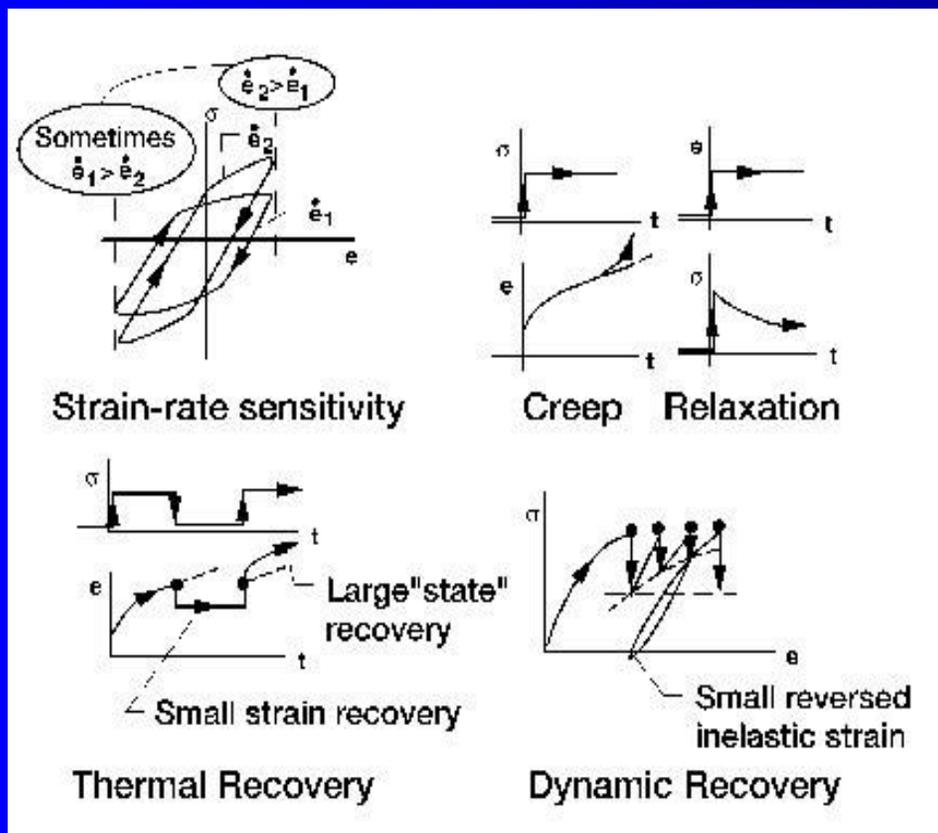
HTHL/SSTO



- Complex Thermomechanical Loading
- Complex Material response requires Time-Dependent/Hereditary Models: Viscoelastic/Viscoplastic
- Comprehensive Characterization (Tensile, Creep, Relaxation) for a variety of material systems



# Important Phenomenological Observations of Behavior of Metals at High Homologous Temperatures ( $T/T_m > 0.3$ )

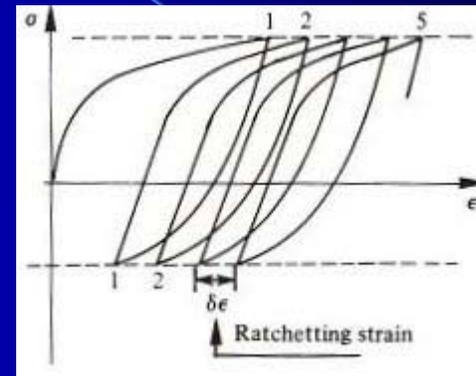
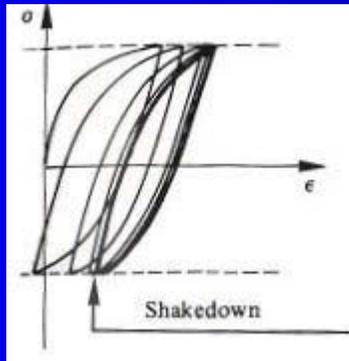


Classic Reason for Introducing Unified Viscoplastic Models (e.g., **GVIPS Class**)

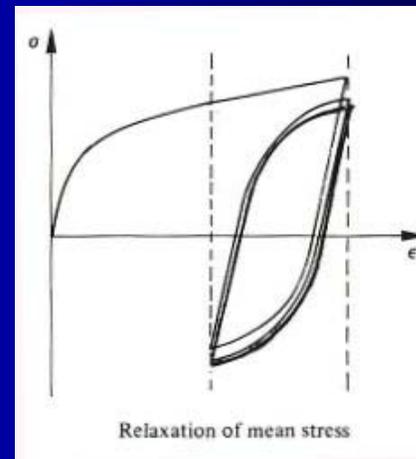
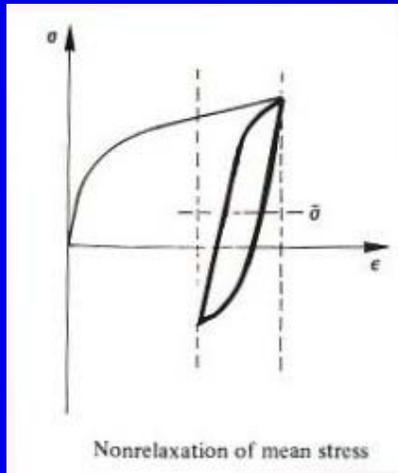
# Important Phenomenological Observations of Behavior of Metals at High Homologous Temperatures ( $T/T_m > 0.3$ )

## Cyclic Behavior

### Stress-controlled



### Strain-controlled



Shakedown Behavior

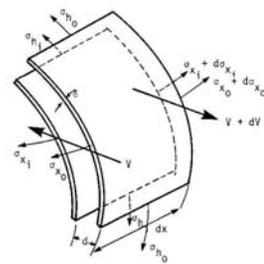
Ratchetting Behavior

# Material Behavior Can Significantly Impact Structural Response (e.g. Recovery Mechanisms)

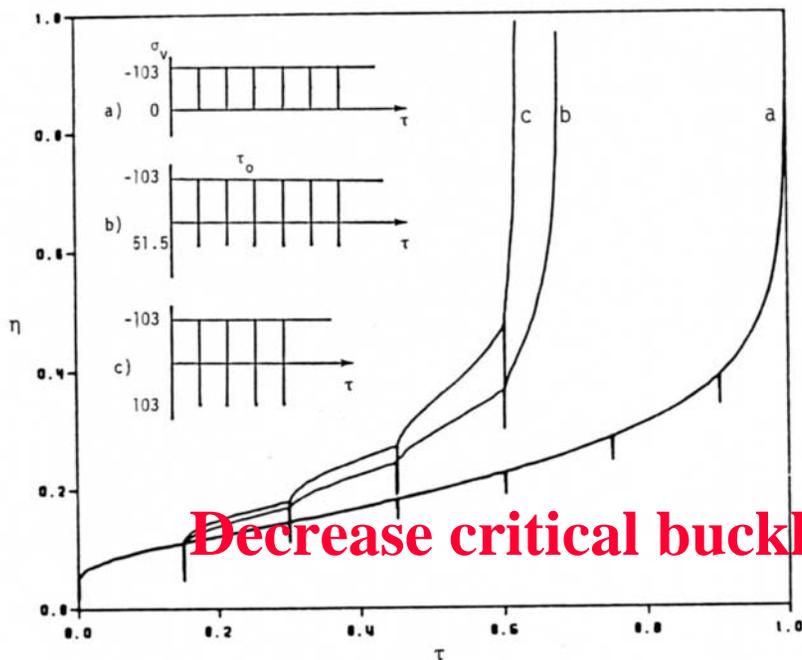
Applied Compressive Stress/Euler Stress = 0.095

Normalized Initial imperfection – 0.01

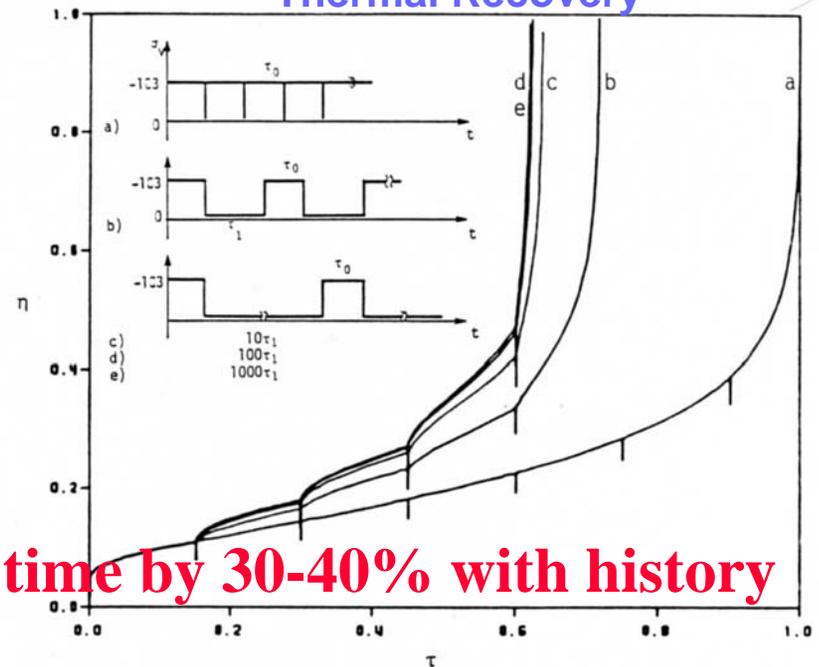
Arnold et al., "Creep Buckling of a Cylindrical Shell Under Variable Loading", Jnl of Eng Mech., ASCE, Vol. 115, No. 5, pp. 1054-1074, 1989.



## Dynamic Recovery



## Thermal Recovery



**Decrease critical buckling time by 30-40% with history**

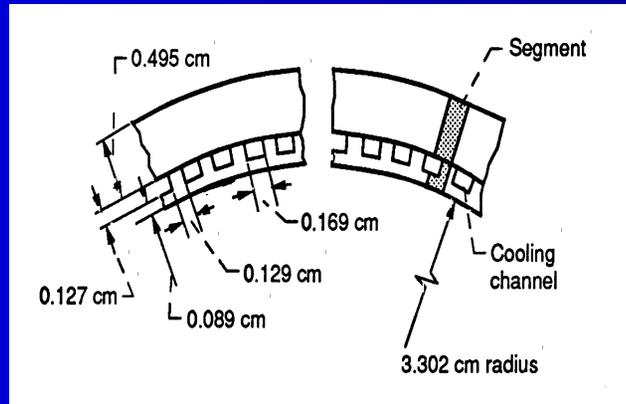
Normalized radial displacement versus normalized time for variable loading histories given in inserts

# Unified Viscoplastic Models Capture Deformation Response in Rocket Engine Nozzle Liners

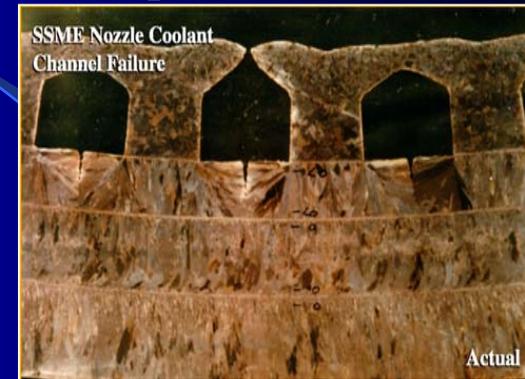
## SSME



## Nozzle Liner Geometry

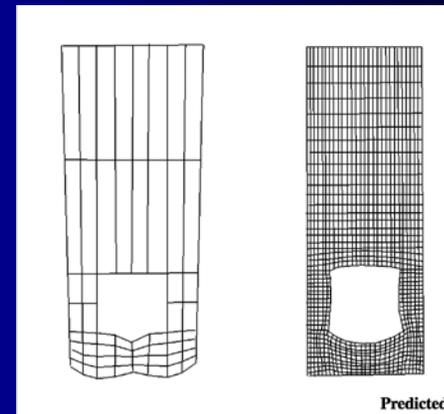


## Experiment (GRC)



## Prediction

Classical (Lockheed)      Unified (GRC)



- Severe thermomechanical loading conditions result in irreversible strains
- Unified viscoplastic models successfully predict the experimentally observed deformation trends
  - Arya and Arnold, AIAA, Vol 30, No. 3, 1992

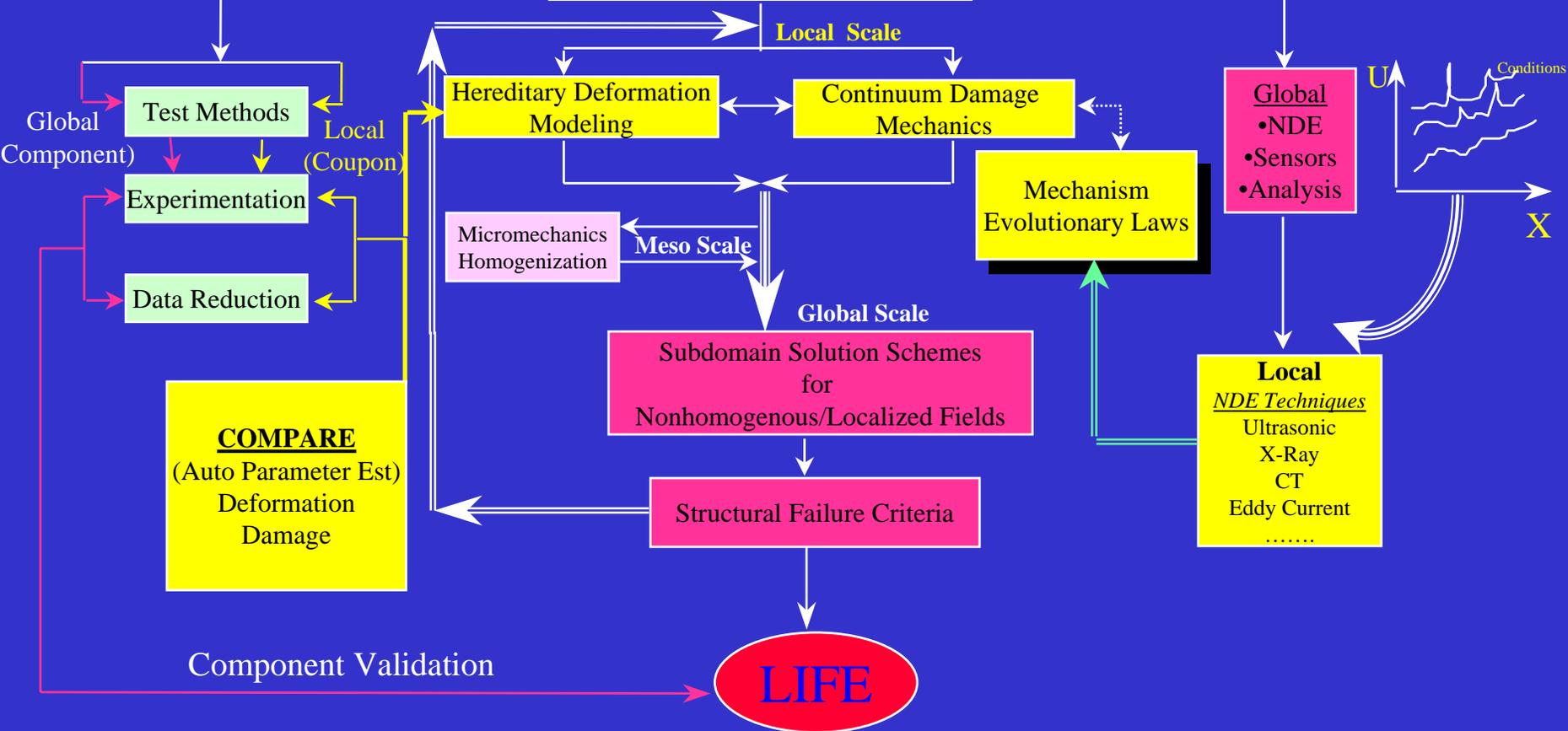


# Multiscale Functional Framework for Deformation and Life Modeling

## Characterization/Validation

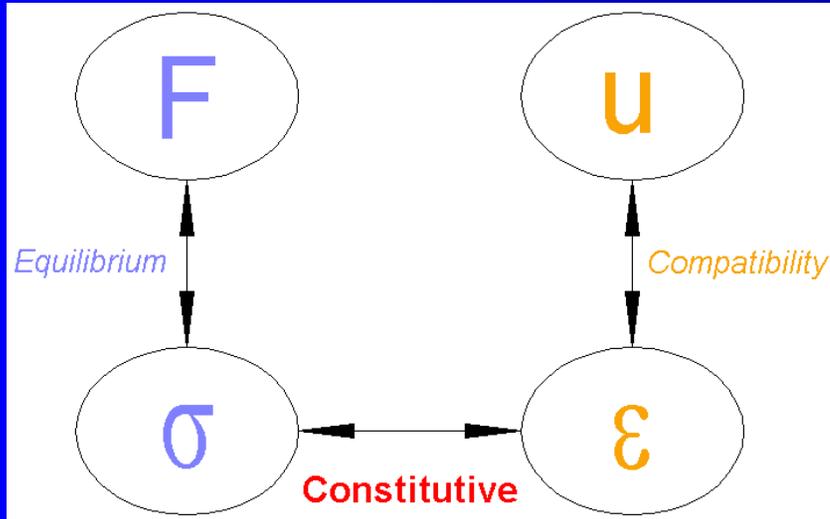
## Structural Analysis

## Detection Techniques



# CONSTITUTIVE MODELING

## Structural Mechanics Problem



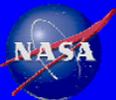
Knowledge of the material's life and constitutive behavior is a prerequisite for assessment of component performance/reliability

Need to concurrently address three important and related areas:

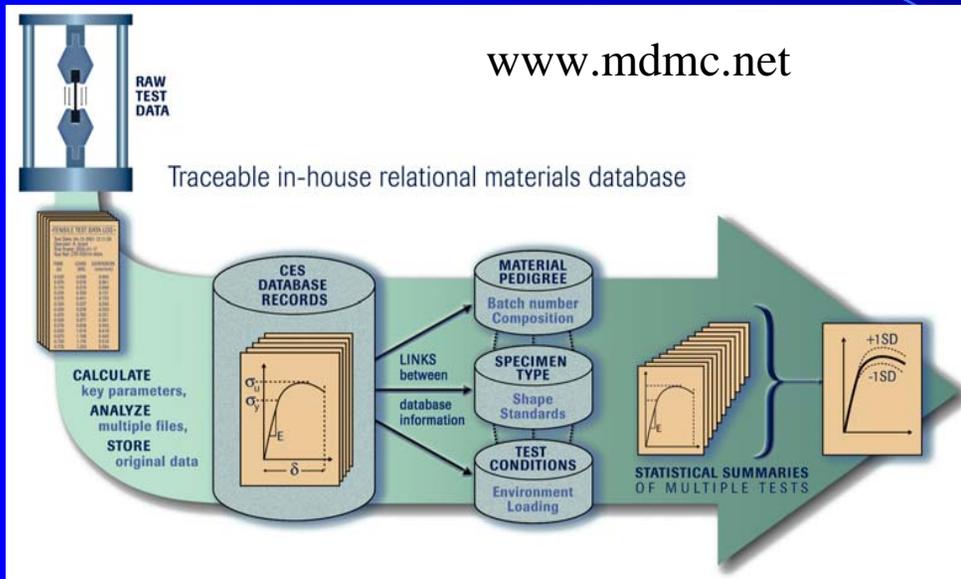
- i) mathematical formulation for the accurate multiaxial representation  
GVIPS Classes
- ii) algorithmic developments for the updating (integrating) of external and internal state variables - FEA User Definable Subroutines
- iii) parameter estimation - COMPARE

This approach allows one to overcome the two major obstacles for practical utilization of sophisticated time-dependent (hereditary) models:

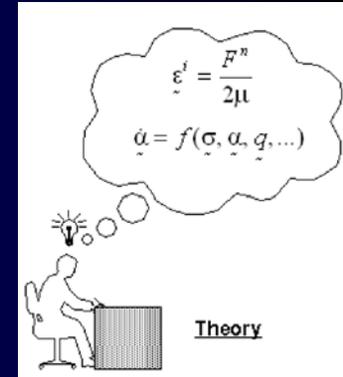
- 1) lack of efficient and robust integration algorithms - FEA Linkage issues
- 2) difficulties associated with characterization of large number of material parameters and appropriate experimental "data content" - COMPARE & sensitivities



# The Desired Vision For Design and Analysis



## Mathematical Characterization Of Material Behavior

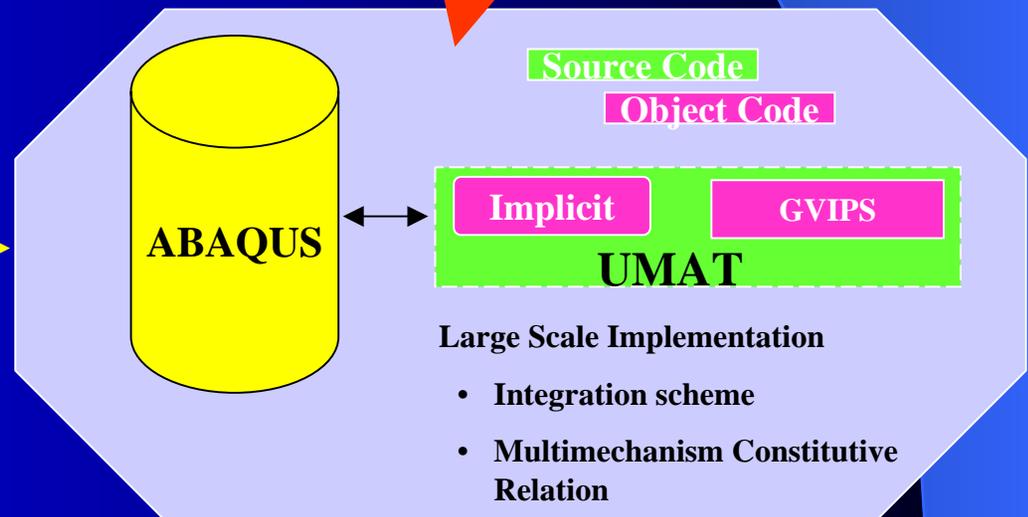


**COMPARE**

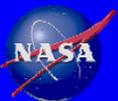
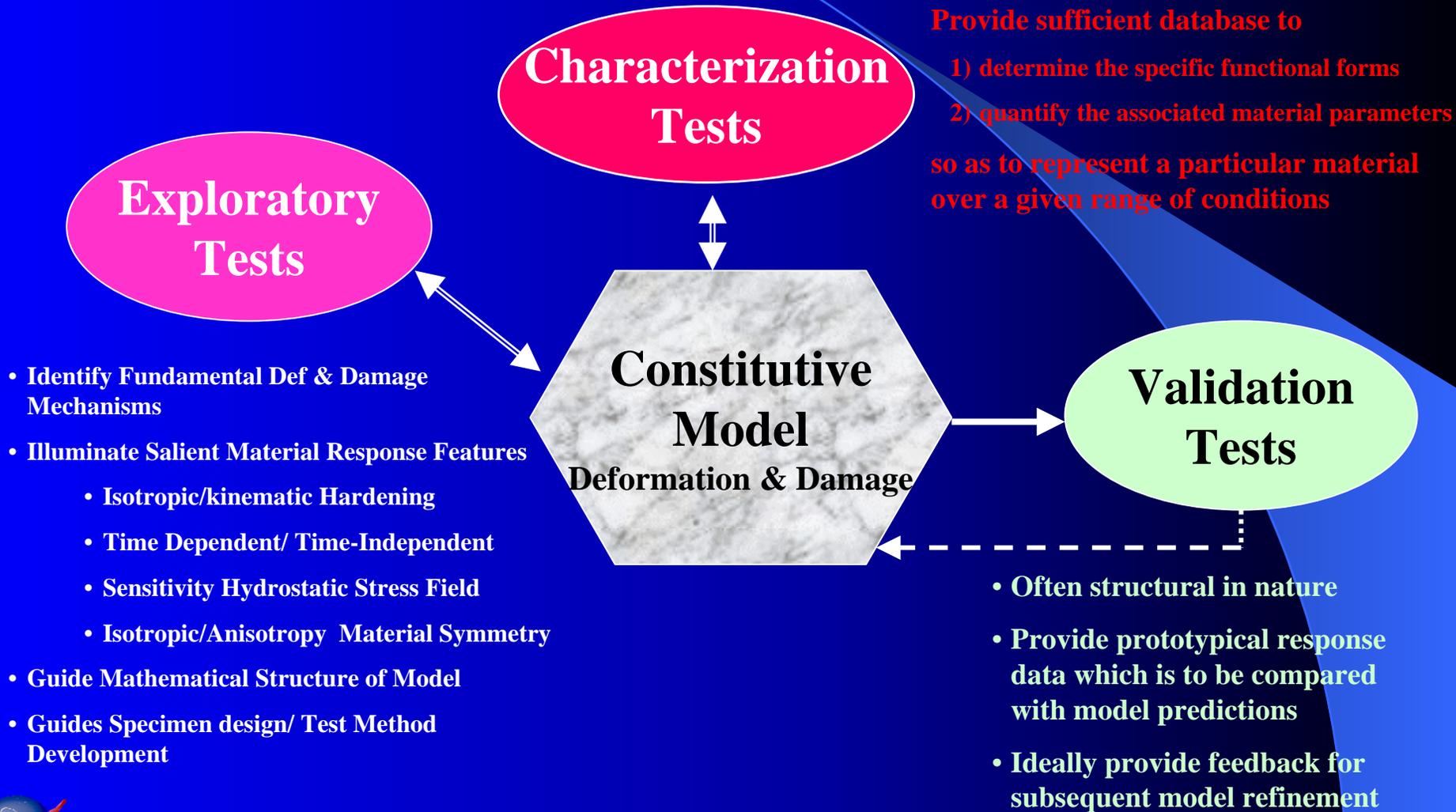
Automatically write required input information



## FEA Analysis of component



# Thermomechanical Testing in Support of Constitutive Model Development



# Experimental Observations

## • Reversibility

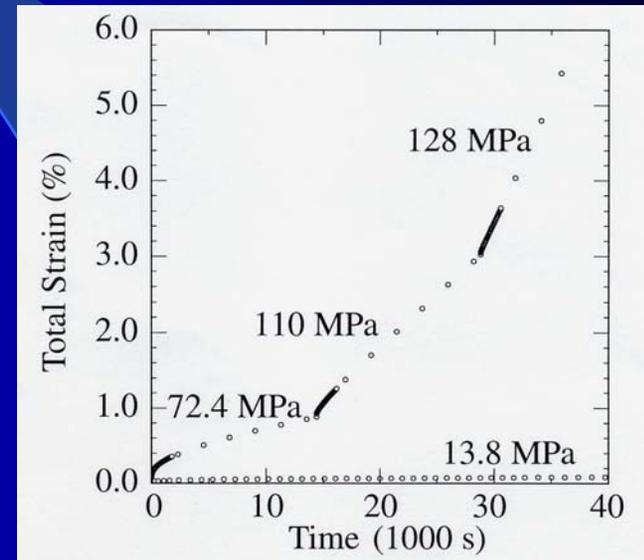
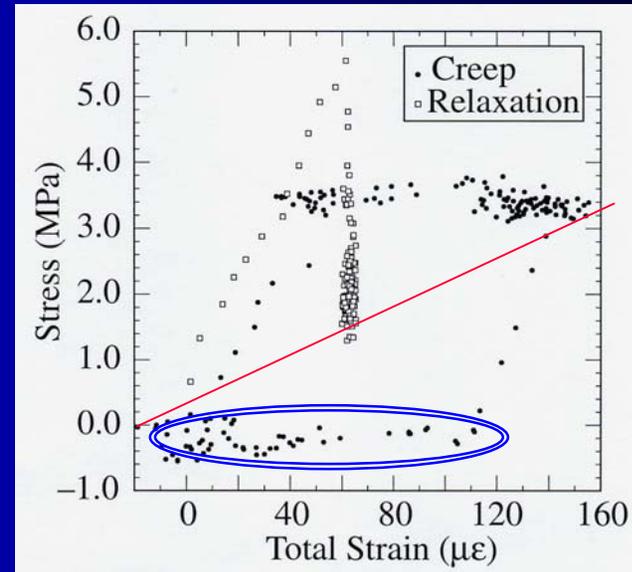
- rate-dependent instantaneous stiffness
- transient creep/relaxation
- **limit equilibrium state**

## • Theoretical demarcation (Exp. Verified)

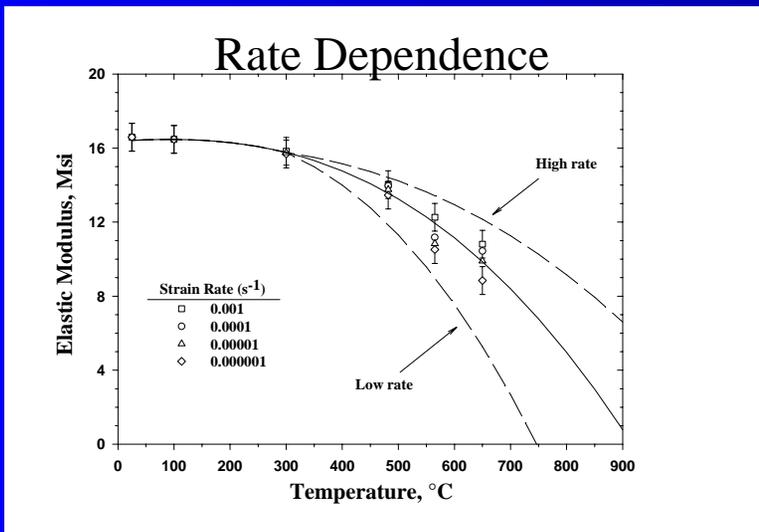
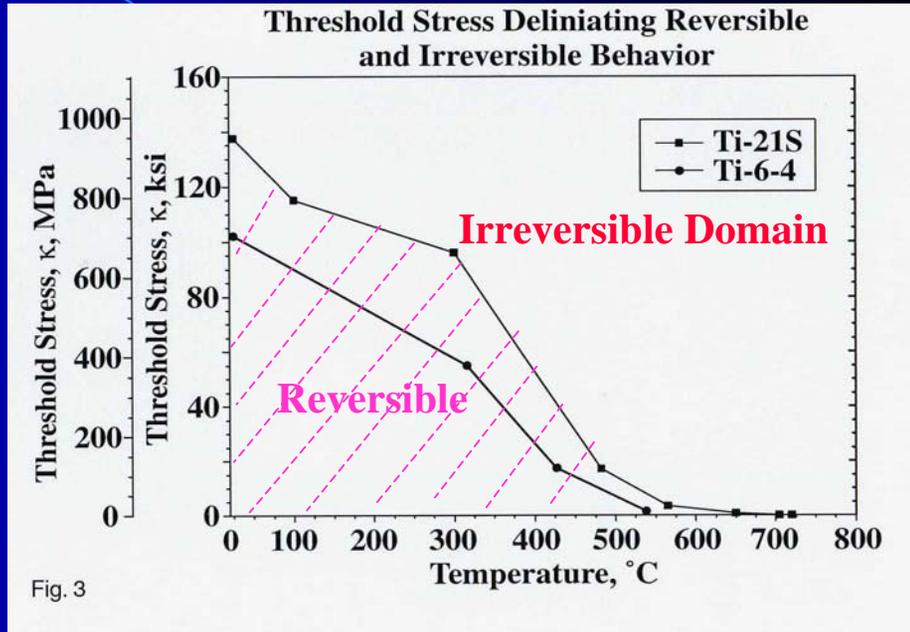
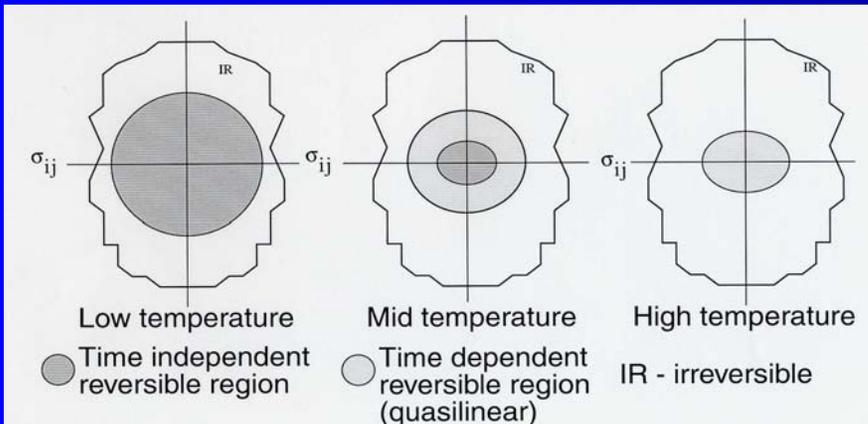
## • Irreversibility

- strain-stress dependent
- nonlinearity
- strain rate dependence
- creep with steady-state
- relaxation with finite residual state
- creep/plasticity interaction
- thermal recovery
- nonlinear kinematic/isotropic hardening

- Anelastic recovery during reversal in both quasilinear and fully developed inelastic regions



# Experiments Indicated Existence of Reversible and Irreversible Threshold Surface



*Experimentally verified for both  
TIMETAL 21S and Ti-6-4*

*GRCop-84 doesn't appear to  
exhibit strong viscoelastic response*



# Theoretical/Computational Motivation

In view of four + decades of active research in the area of inelastic behavior modeling, the **need** still exists for an:

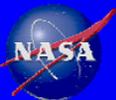
*Accurate representation of material response details over an extensive domain of time, stress, temperature, loading conditions ...*

## Assessment

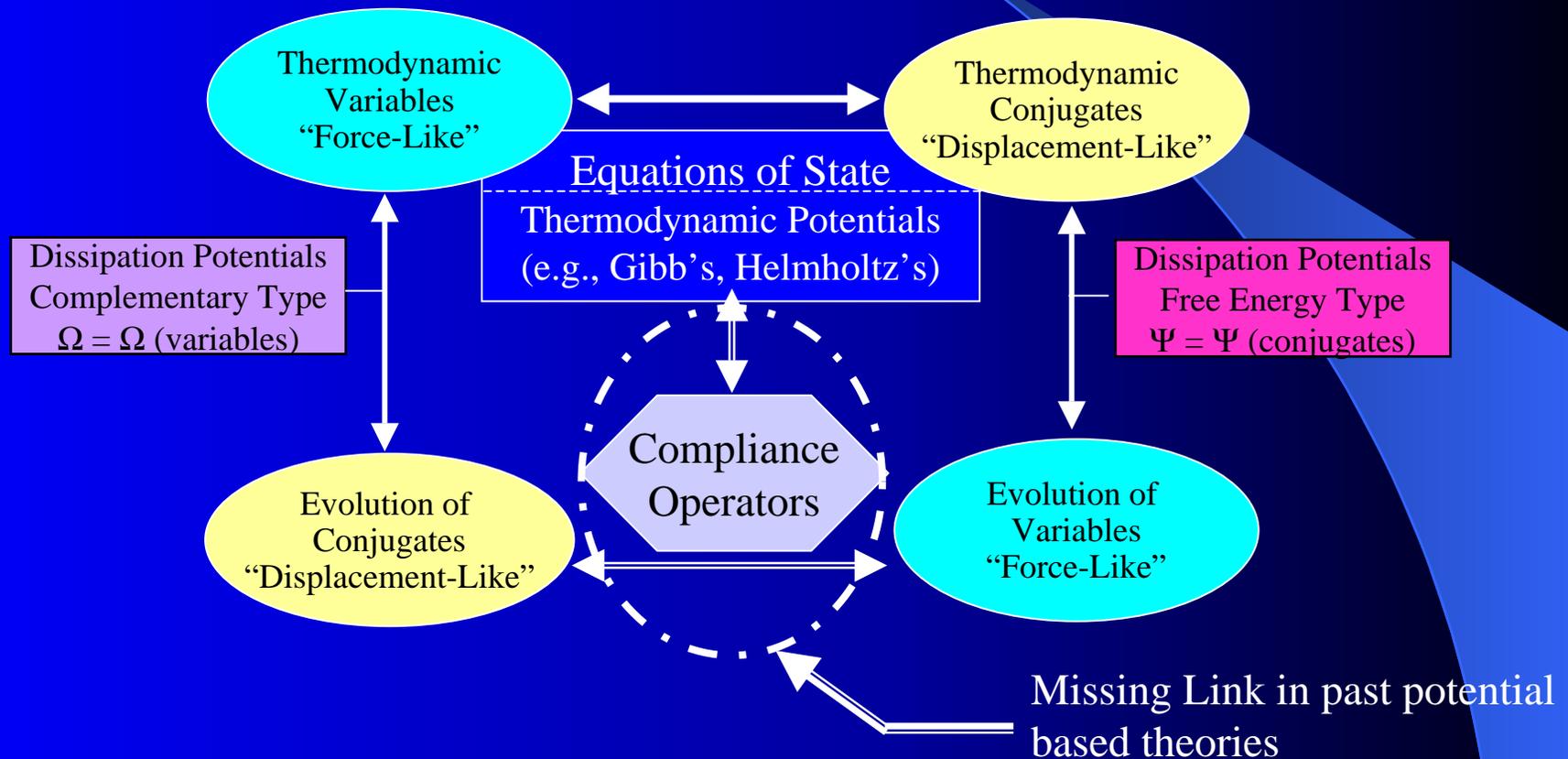
### Technical

### Practical Implication

Non-associative - Nonsymmetric Tangent Stiffness  - Coupled system of Stiff Diff. Eq.	⇒ Non-uniqueness of solution ⇒ Implementation into large scale FEA codes problematic ⇒ Difficult to integrate
Numerous nonphysical material parameters	⇒ Requires expertise to characterize model
Single-mechanism models	⇒ Qualitatively capable, yet quantitatively limited in response spectrum

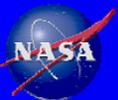


# Utilize Concept of Thermodynamic Internal State Variables to Obtain Constitutive Equations

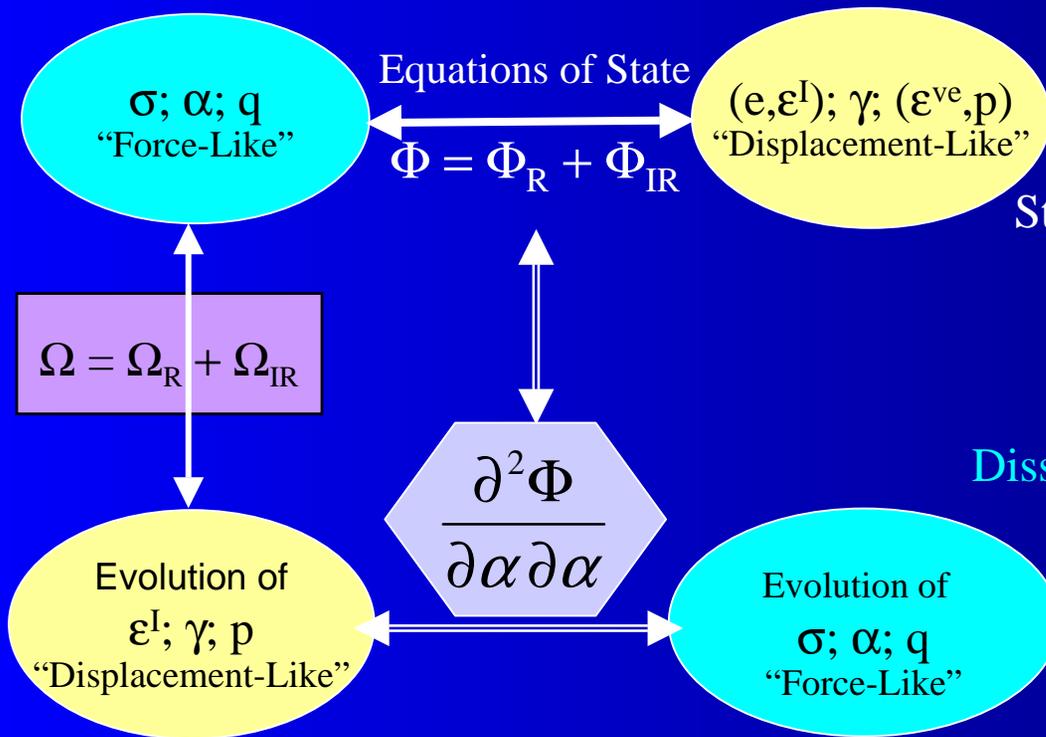


# Advantages and Attributes of Potential Formulation

- Provides a consistent framework for deformation and damage modeling
  - Nonisothermal and/or anisotropic extension straight forward
  - Nonproportional loading histories automatic
  - Automatic satisfaction of the Dissipation Inequity of Thermodynamics
- Eliminates the “ad-hoc” nature of model development
- Provides sufficiently general variational structure.
- Constitutes cornerstone of regularity and bounding (or limit) theorems in plasticity and viscoplasticity.
- Lends itself to robust numerical implementation



# Physical Mechanisms Underlying The Partitioning of Energy : Complementary Type



**Total** = Stored + Dissipated

$$\sigma e = \Phi + \Omega$$

Stored ( $\Phi$ ) = Reversible + Irreversible

Lattice Distortion    Dislocation Pile-up

Reflects change in microstructure

Dissipation ( $\Omega$ ) = Reversible + Irreversible

Dislocation bowing    Deformation & Thermally driven Mechanism

Reflects mobility/rate of evolution in microstructure

**Irreversible** =  $\Omega_1$  (deformation) +  $\Omega_2$  (diffusional; mass/vacancy)

Glide/plastic Slip

- Thermal recovery
- Dislocation/boundary interaction
- Formation of cell structure

# General Multimechanism Hereditary Behavior Model of the GVIPS Class

Reversible

Irreversible

$\Phi = \Phi_R + \Phi_{IR} \quad \Omega = \Omega_R + \Omega_{IR}$

$\Phi_R = \Phi_R(\sigma, q^{(a)}) = \frac{1}{2} \sigma_s : E^{-1} : \sigma_s + \frac{1}{2} \sum_{a=1}^M q^{(a)} : [M^{(a)}]^{-1} : q^{(a)} + \sum_{a=1}^M q^{(a)} : p^{(a)}$

$\Phi_{IR} = \Phi_{IR}(\sigma, \alpha^{(b)}) = \sigma : \epsilon^{vp} + \sum_{b=1}^N \bar{H}_{(b)}(\alpha^{(b)})$

$\Omega_R = \frac{1}{2} \sum_{a=1}^M q^{(a)} : [\eta^{(a)}]^{-1} : q^{(a)}$

$\Omega_{IR} = \Omega_1(\sigma - \alpha) + \sum_{b=1}^N \Omega_2^{(b)}(\alpha^{(b)})$

**Eqs. of state:**  $\epsilon^{ve} = \left( \frac{-\partial \Phi_R}{\partial \sigma_s} \right)$

**Flow law:**  $\dot{p}^{(a)} = \frac{\partial \Omega_R}{\partial q^{(a)}}; \quad a = 1, 2, \dots, M$

**Flow law:**  $\dot{\epsilon}^{vp} = \frac{\partial \Omega}{\partial \sigma} = \frac{\partial \Omega_1}{\partial \sigma}$

**Evolutionary law:**  $\dot{\gamma}^{(b)} = \frac{-\partial \Omega_{IR}}{\partial \alpha^{(b)}}; \quad b = 1, 2, \dots, N$

**Internal State Rate Eqs.:**  $\dot{\alpha}^{(b)} = L \dot{\gamma}^{(b)}; \quad L = \left[ \frac{-\partial^2 \Phi_{IR}}{\partial \alpha^{(b)} \partial \alpha^{(b)}} \right]^{-1}$

# Specific Choice of Energy Potentials and Material Functional Forms

$$\Phi_R = \Phi_R(\sigma_{ij}, \mathbf{q}_{ij}^{(a)}) = \frac{1}{2}(\sigma_s)_{ij} \mathbf{E}_{ijkl}^{-1}(\sigma_s)_{kl} + \frac{1}{2} \sum_{a=1}^M \mathbf{q}_{ij}^{(a)} [\mathbf{M}_{ijkl}^{(a)}]^{-1} \mathbf{q}_{kl}^{(a)} + \sum_{a=1}^M \mathbf{q}_{ij}^{(a)} \mathbf{P}_{ij}^{(a)}$$

$$\Phi_{IR} = \Phi_{IR}(\sigma_{ij}, \alpha_{ij}^{(b)}) = \sigma_{ij} \varepsilon_{ij}^{up} + \sum_{b=1}^N \bar{H}_{(b)}(G^{(b)})$$

Stored Energy

and

$$\Omega_R = \frac{1}{2} \sum_{a=1}^M \mathbf{q}_{ij}^{(a)} [\boldsymbol{\eta}_{ijkl}^{(a)}]^{-1} \mathbf{q}_{kl}^{(a)}$$

$$\Omega_{IR} = \Omega_1(F) + \sum_{b=1}^N \Omega_2^{(b)}(G^{(b)})$$

Dissipation

where

$$F = \frac{1}{2\kappa^2} (\sigma_{ij} - \alpha_{ij}) \mathcal{M}_{ijkl} (\sigma_{kl} - \alpha_{kl}) - 1$$

$$G^{(b)} = \frac{1}{2\kappa_{(b)}^2} (\alpha_{ij}^{(b)} \mathcal{M}_{ijkl} \alpha_{kl}^{(b)})$$

and the specific functions :

$$\Omega_1(F) = \int \frac{\kappa^2 F^{\alpha}}{2\mu} dF$$

$$\Omega_2^{(b)}(G^{(b)}) = \kappa_{(b)}^2 \int \frac{r(G^{(b)})}{h(G^{(b)})} dG^{(b)}$$

$$\bar{H}_{(b)} = \kappa_{(b)}^2 \int \frac{1}{h(G^{(b)})} dG^{(b)}$$

$$h_{\text{const}}(G^{(b)}) = \frac{H_{(b)}}{[G^{(b)}]^{\beta_{(b)}}}$$

$$r(G^{(b)}) = R_{(b)} [G^{(b)}]^{m_{(b)}}$$

$$h_{\text{sat}}(G^{(b)}) = H_{(b)} \left\langle 1 - \sqrt{G^{(b)}} \right\rangle^{\beta_{(b)}}$$



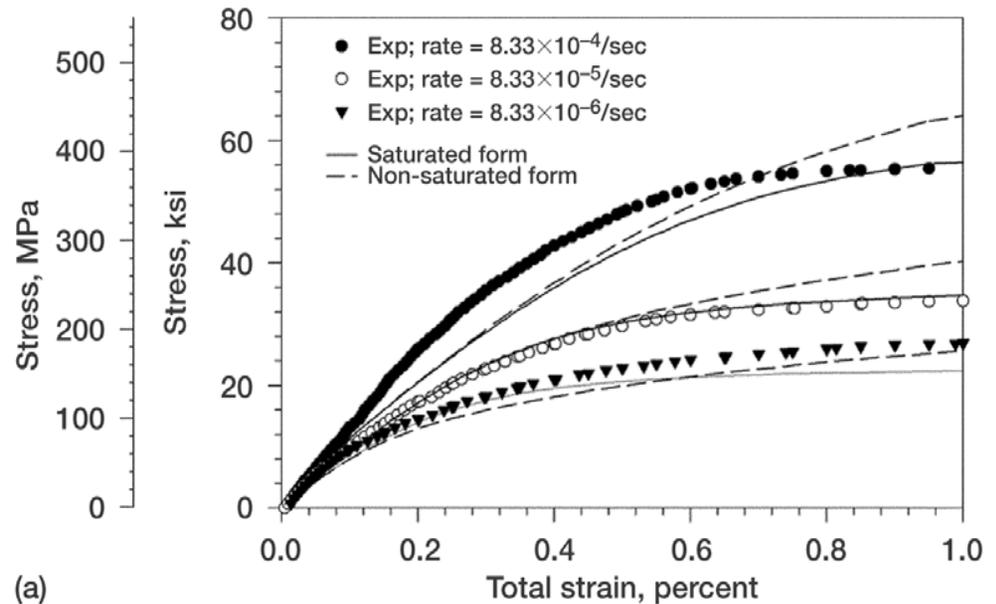
# Results Illustrating Recent Improvements Made to the Hardening Functional Form in GVIPS Model

Previous Non-saturating  
 $g(G) = H / G^\beta$

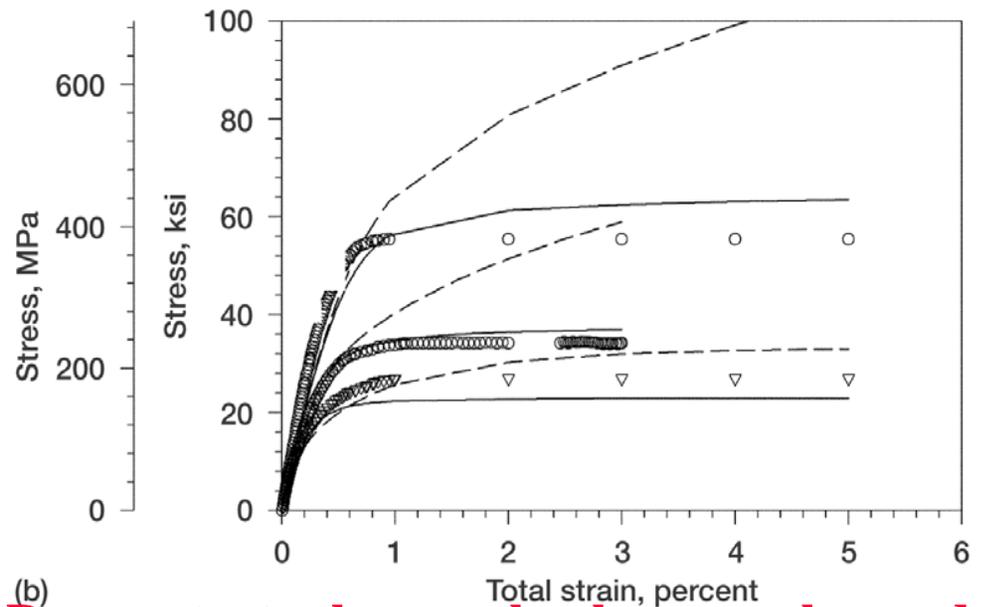
Current Saturating Form  
 $g(G) = H(1-G)^\beta$

$$G = [1/2(\alpha_{ij} \alpha_{ij}) / K^2_{(b)}]^{0.5}$$

TIMETAL 21S: 650°C  
 Strain Controlled Tensile  
 Single Mechanism



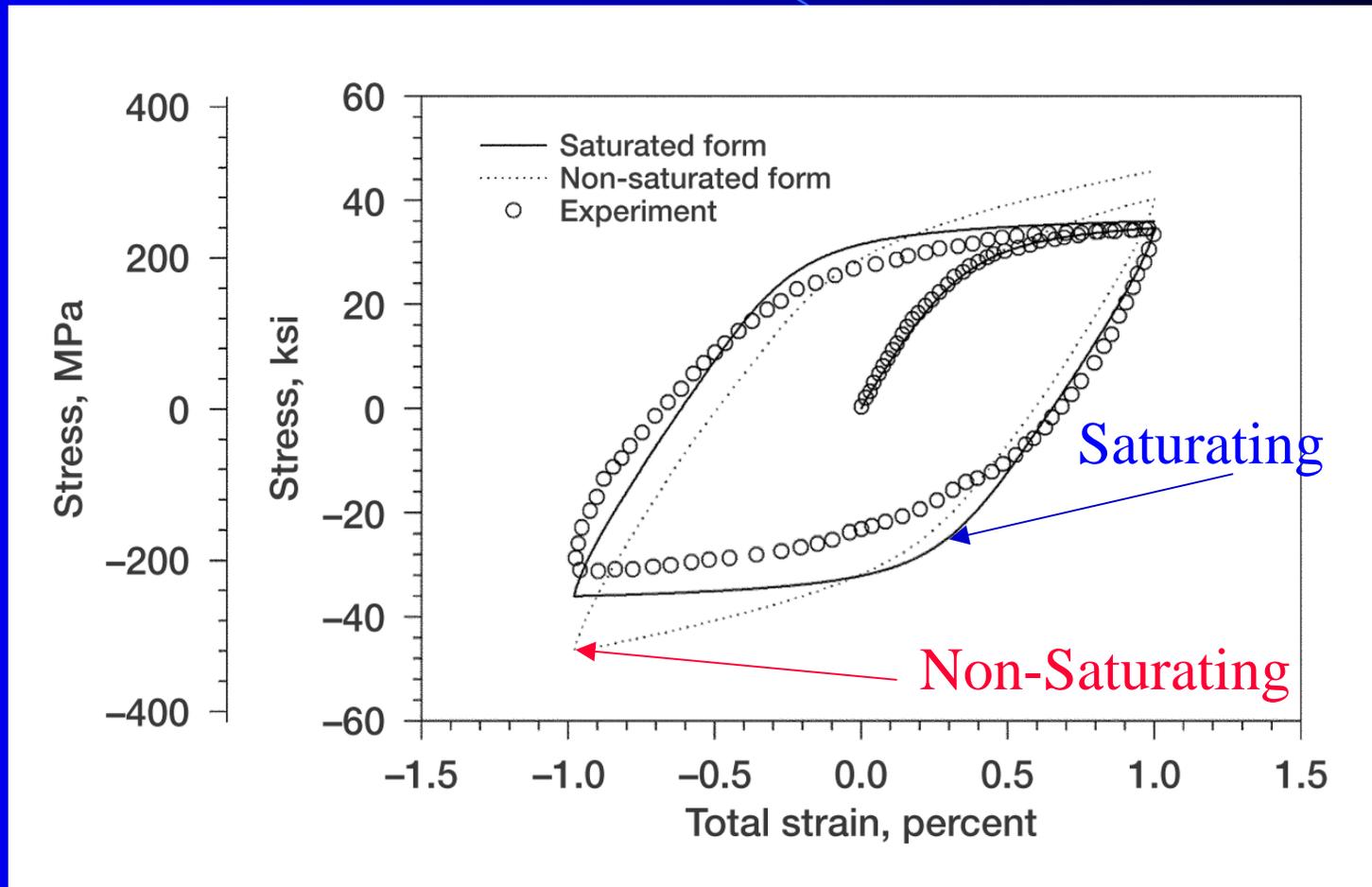
(a)



(b)

**Demonstrates how scale-abuse can be used**

# Comparison of Specific Hardening Forms Under Cyclic Loading



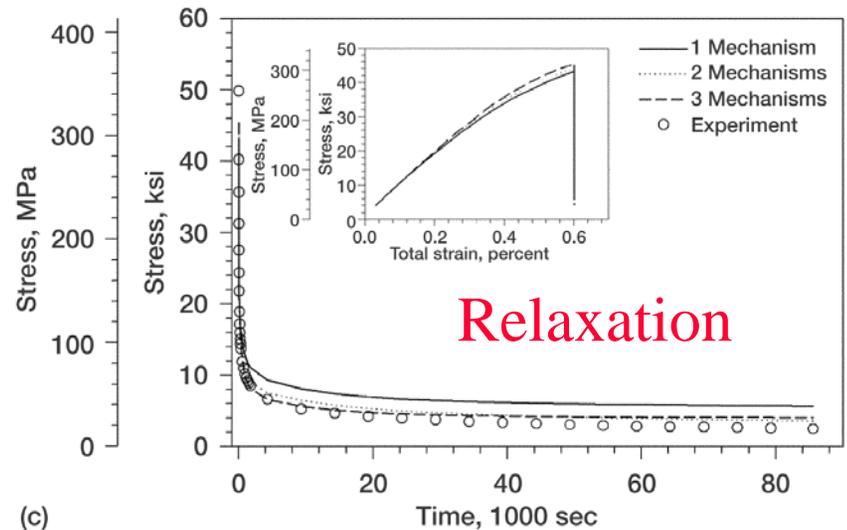
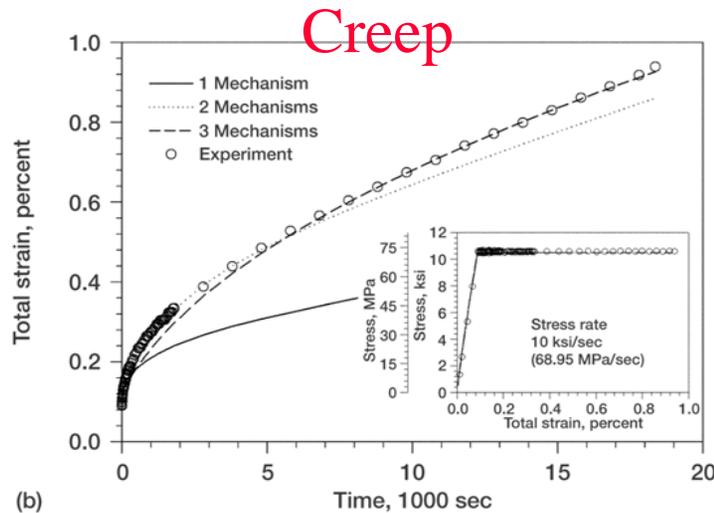
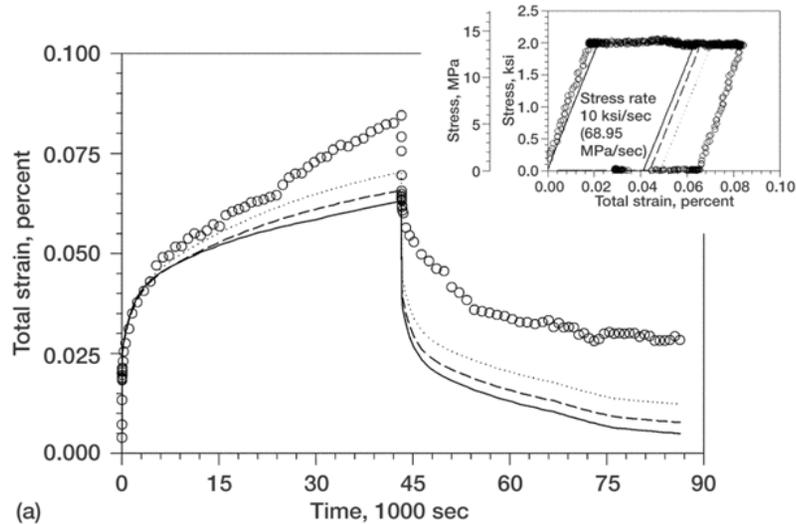
TIMETAL 21S: 650°C

Strain Controlled



# New Saturating Form Does Not Adversely Impact Ability to Represent Creep/Relaxation

- But need at least two mechanisms to capture both creep and relaxation well



# ***Robust Integration Scheme Key For Efficient Inelastic Finite Element Analysis***

Common approaches for integration of rate equations:

1) Non-Iterative: explicit; semi-implicit

No local iterations  less overhead  
stability problems

2) Iterative: fully-implicit

Requires local iterations  additional overhead

Unconditional stability

Consistent Tangent Stiffness

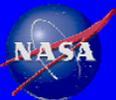
Quadratic Convergence of global  
Newton-Raphson Iterations

**Selected:**

**Backward Euler with Line Search**

## Advantages of Implementation

- Directly applicable for 3-D and sub-space loading(plane strain, axisymmetric, etc)
- Generalized Material Symmetry Operators (which influence flow, hardening, recovery, relaxation spectrum, etc.)
- Efficiency (through explicit algorithmic tangent stiffness)
- Robustness (through “slack” line search)

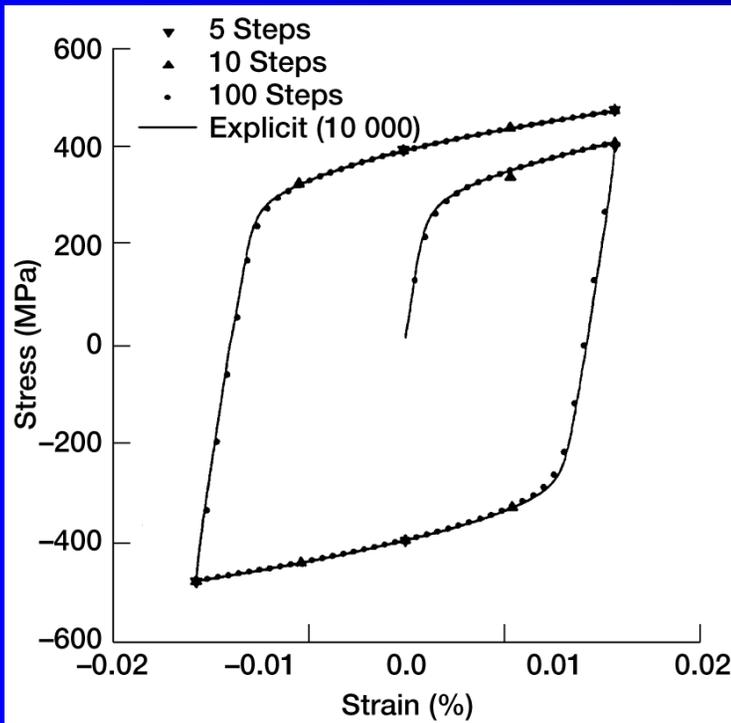


# Results Illustrating the Efficiency of The Numerical Implementation of GVIPS

## Backward Euler with Line Search

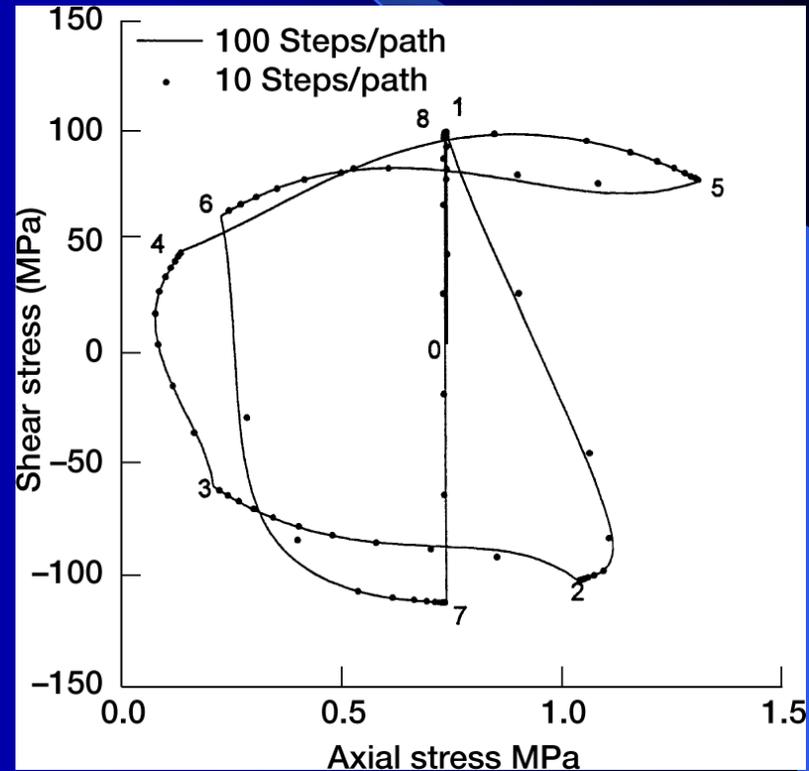
W/Kanthal ,  $\dot{\epsilon}=2 \times 10^{-3} / s$  ,  $\epsilon_{max}=0.0144$

method	number of load steps	CPU time	GIT	LIT
explicit	10,000	180.0	3	0
implicit	100	5.0	2	4
implicit	10	1.05	4	10
implicit	5	1 ( 54 s )	10	20



Under cyclic conditions

\*\*Explicit Failed



Under nonproportional loading conditions



# Key to Accurate Characterization of GVIPS Involves Sufficient “Data Content”

## Viscoelastic Material Parameters

2+2M number, i.e.,  $E_s$ ,  $\nu$ ,  $(M_{(a)}, \rho_{(a)})$

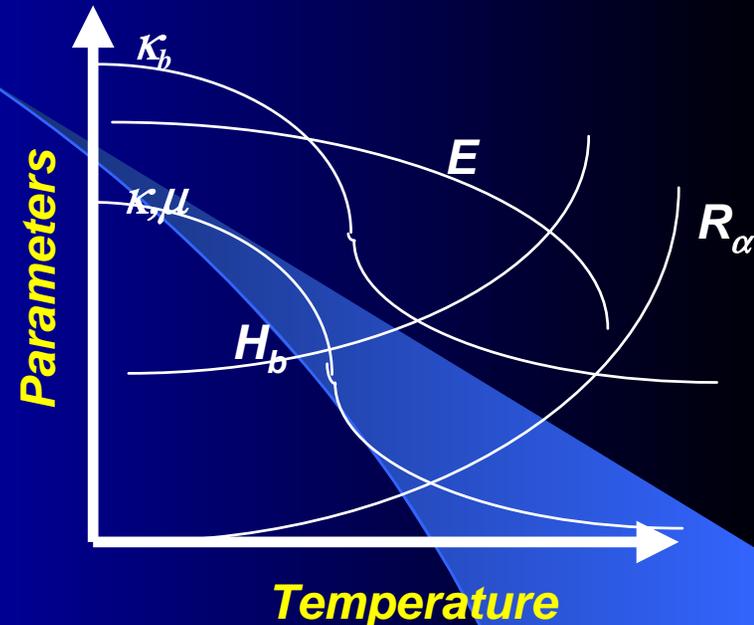
## Viscoplastic Material Parameters

- Flow  $\kappa$ ,  $\mu$ ,  $n$
- Hardening  $H_b$ ,  $K_b$  and  $\beta$ ,
- Recovery:  $R_b$  and  $m_b$

3 + 5N irreversible material constants

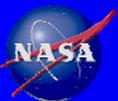
## Types of Experimental Tests

- Strain controlled Tensile Tests (multiple rates)
- Creep Test (Monotonic and/or step)
- Relaxation (Monotonic and/or step)
- Cyclic Tests (Fully reversed, ratcheting)
- Biaxial Tests (tensile, creep, relaxation, cyclic)

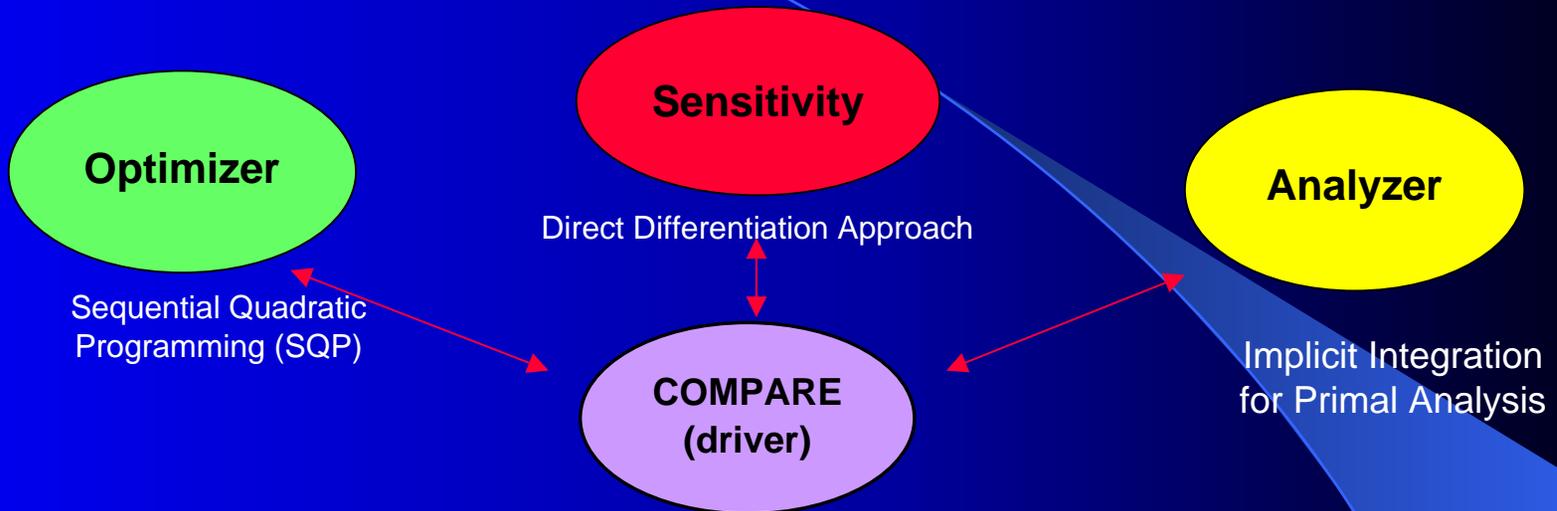


Desire a **mixture** (rather than numerous of one type) **of tests** at numerous temperatures

**Quality vs. Quantity**



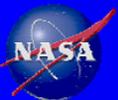
# COMPARE CORE



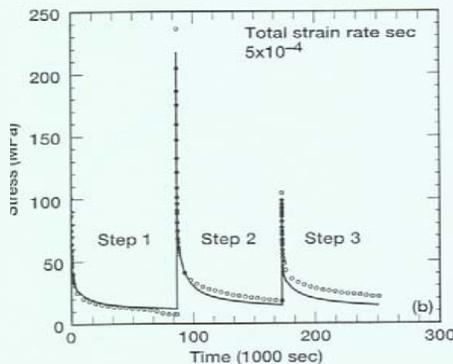
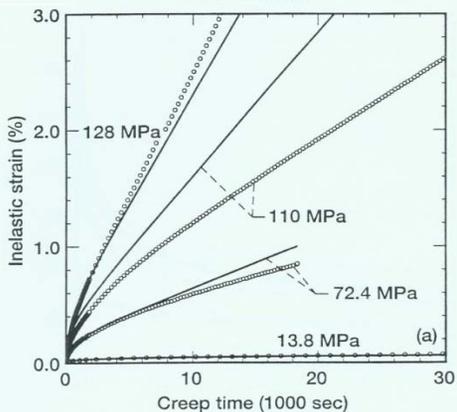
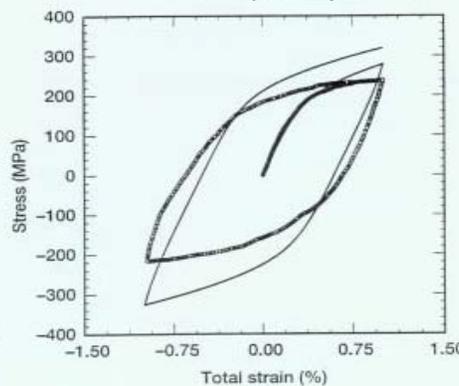
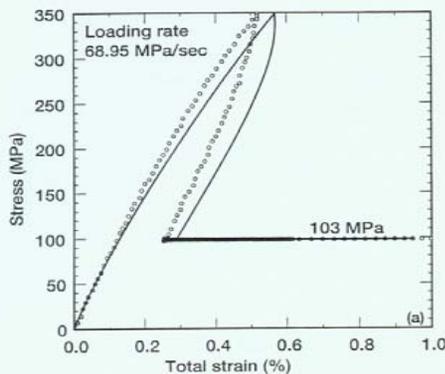
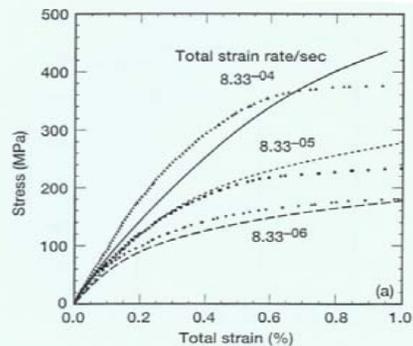
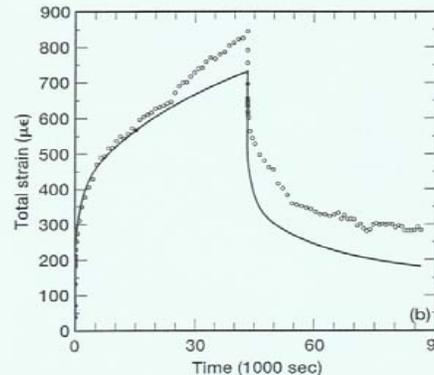
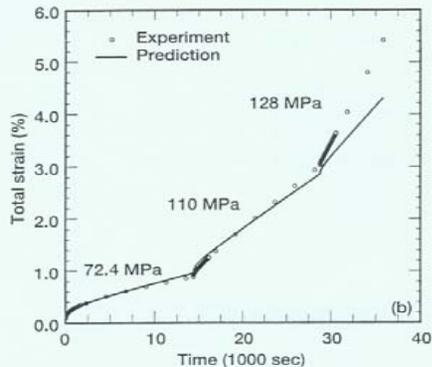
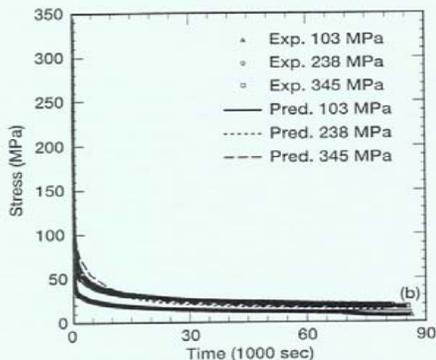
- Identify active/passive variables for a test
- Scaling design variables and objective function
- Formulating a single design optimization problem weighted objective function.  
Constraints  
sensitivities

**Results**

- Final Optimum Material Parameters
- Combined & Individual Error Functions



# Comprehensive Characterization of The Deformation Response of TIMETAL21S



**“DATA CONTENT” IS HUGE ISSUE**

## Wide Range of Application

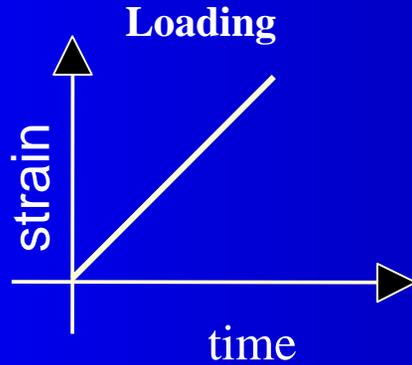
Stress: 1 → 60 Ksi

Time: 2 → 90000 sec

Temp: 650 C

Loading Rates:  $10^{-2} \rightarrow 10^{-10}$

# Characterization of IN738LC @ 850 °C

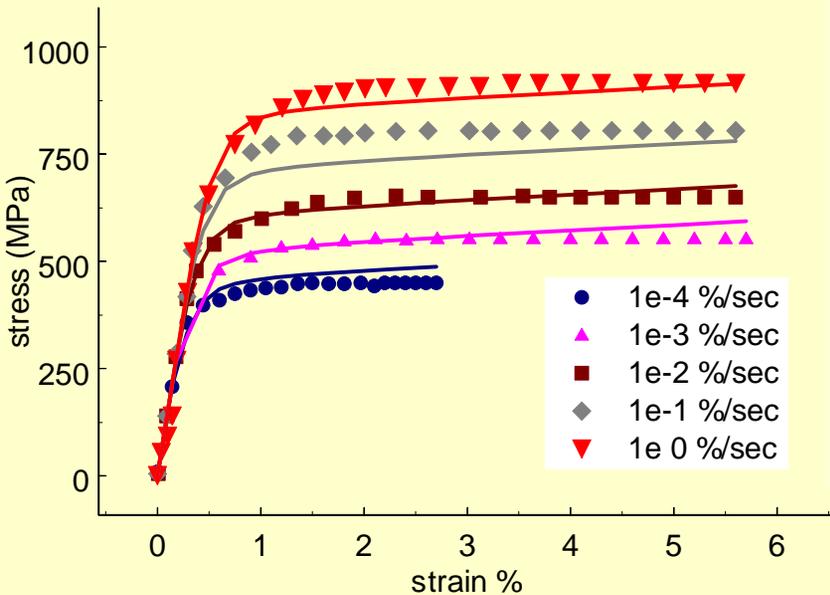


## Elastic + 4 Viscoplastic Mechanisms

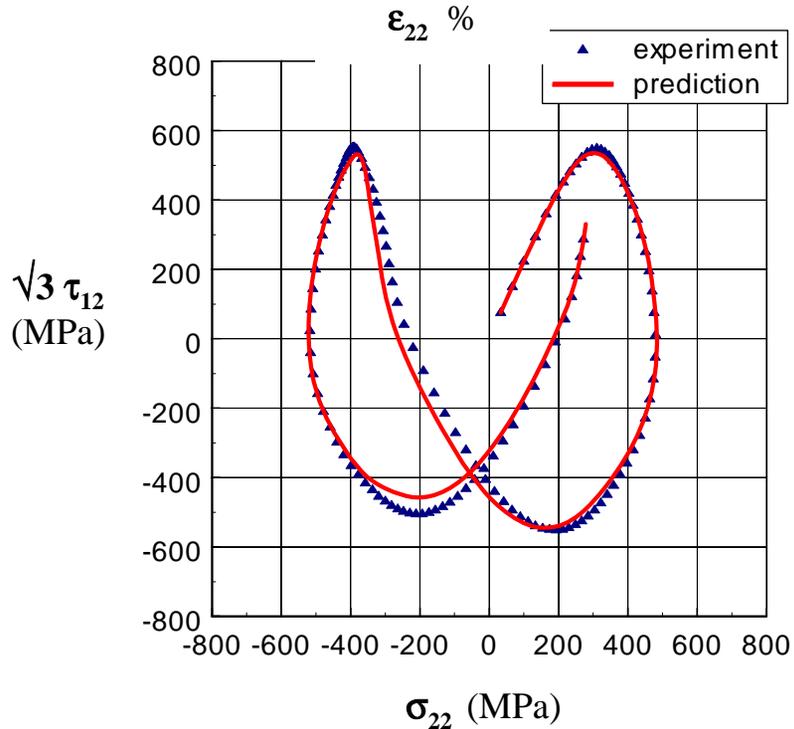
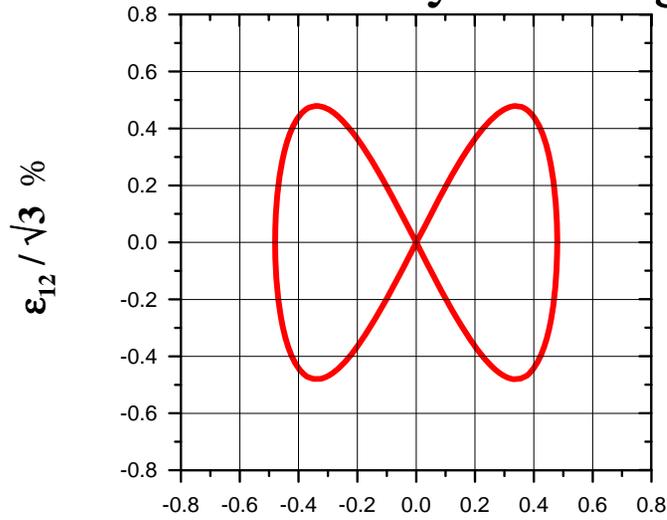
Final characterized parameters using four Viscoplastic mechanisms for IN738LC @850<sup>0</sup> C

Material Parameter	Units	Value	Material Parameter	Units	Value
E	MPa	$1.5 \times 10^5$	$\beta_1$	-	1 (6)*
$\nu$	-	0.33	$\beta_2$	-	1 (6)*
$\kappa$	MPa	0.1	$\beta_3$	-	1 (6)*
$\kappa_1$	MPa	61.43	$\beta_4$	-	1 (6)*
$\kappa_2$	MPa	64.37	$R_1$	1/s	$1.0 \times 10^{-21}$
$\kappa_3$	MPa	62.30	$R_2$	1/s	$1.0 \times 10^{-21}$
$\kappa_4$	MPa	75.08	$R_3$	1/s	$1.0 \times 10^{-21}$
n	-	1.486	$R_4$	1/s	$1.0 \times 10^{-21}$
$\mu$	MPa -s	$3.79 \times 10^{14}$	$H_1$	MPa	$4.6 \times 10^4$
$m_1$	-	0.001	$H_2$	MPa	$5.13 \times 10^4$
$m_2$	-	0.001	$H_3$	MPa	$8.33 \times 10^7$
$m_3$	-	0.001	$H_4$	MPa	$9.458 \times 10^7$
$m_4$	-	0.001			

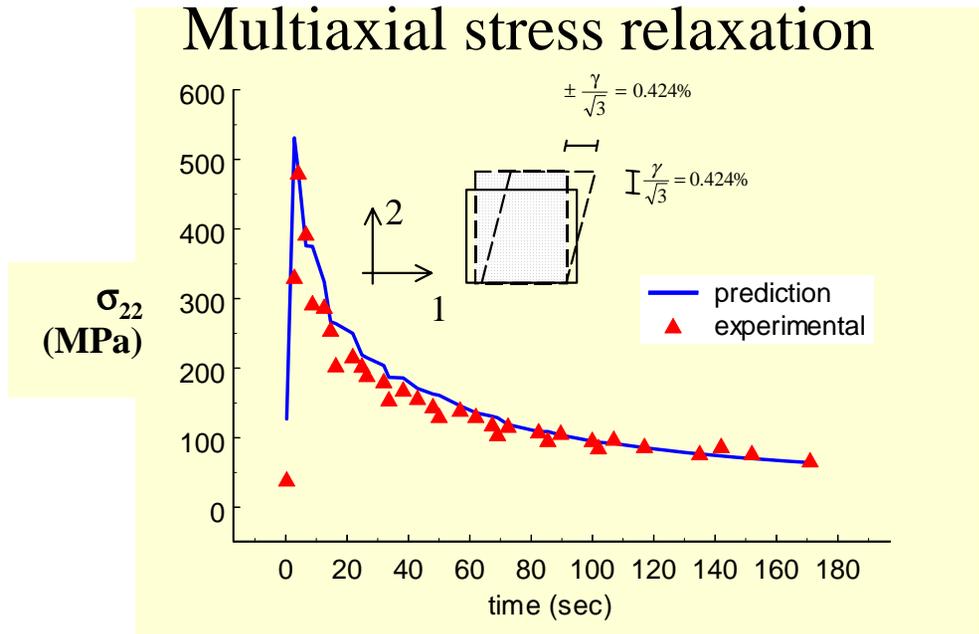
\* the value between parentheses was determined in the FE simulation of the experiment



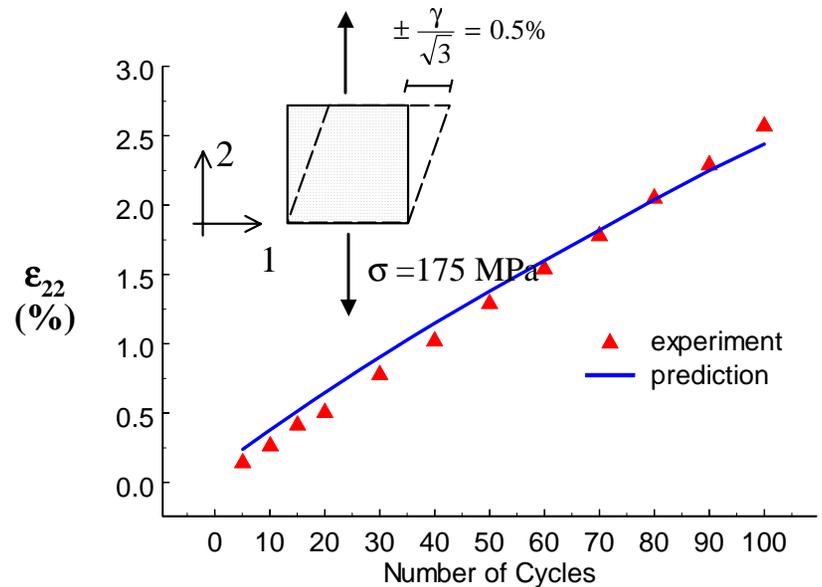
# Butterfly Loading



# Multiaxial stress relaxation

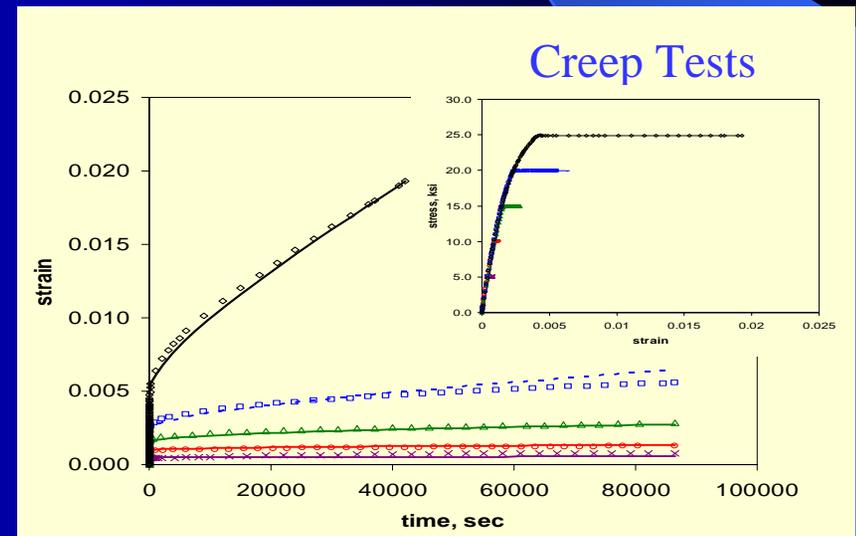
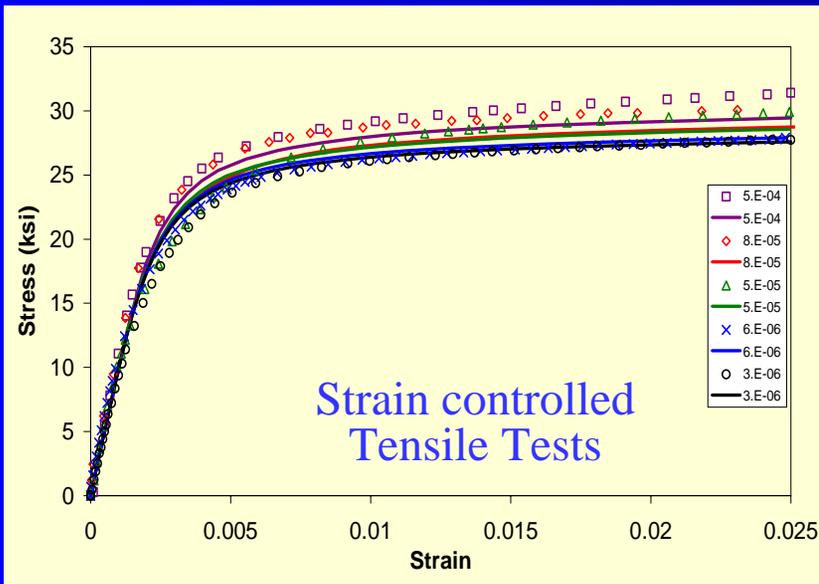
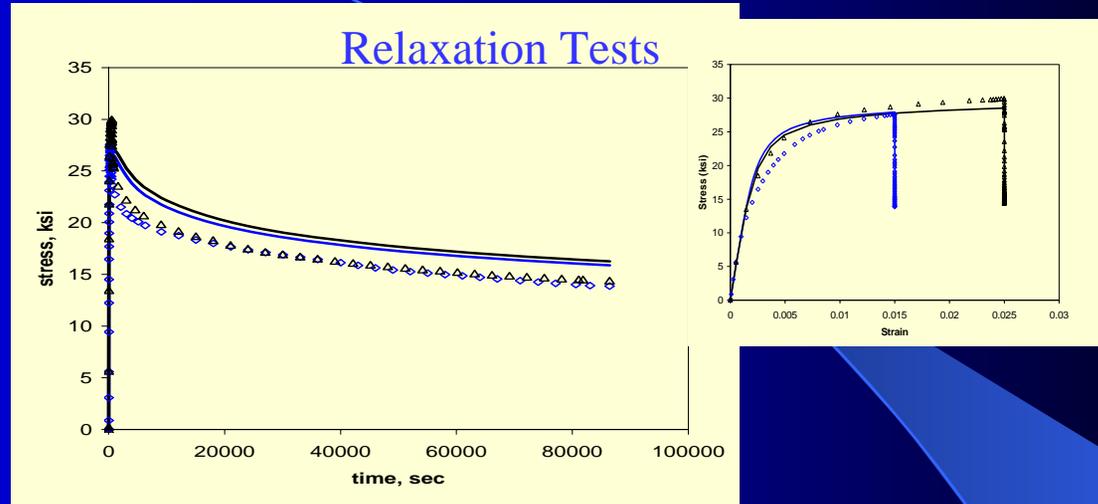


# Multiaxial ratchetting



# Correlation of GRCop-84 Utilizing Multimechanism GVIPS Model

1 VE mechanisms  
4 VP mechanisms



# Structural Verification Testing

- Ideally should provide feedback for subsequent model refinement
- Provide prototypical response data which is to be **compared** with model predictions



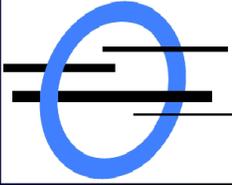
## Consequently:

- Need accurate temperature, strain and load information at a variety of locations - **required for any true validation**
- Number of cycles to failure (alone) not enough
- Instrumentation incredibly challenging (sever environment)

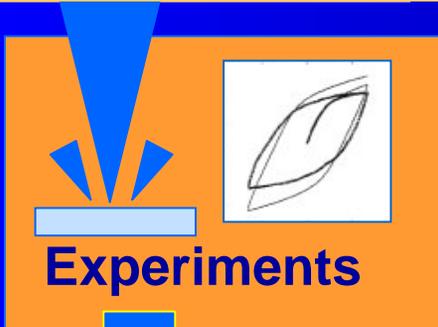
# Summary of Advances in Material Modeling (Synergistic Technology)

- Generalized, Fully Associative, Multimechanism, Viscoelastoplastic Model Available
  - Reversible/Irreversible Regimes
  - Spanning wide time, stress, temperature spectrum
  - Nonlinear Hardening with Saturation
  - Ability to capture ratcheting
  - Stiffness and/or Strength Reduction
- Automated Material Model Characterization
  - via **COMPARE**
  - **Materials thus far:**
    - Ni based; Cu based; Ti
    - MMC and PMC
- Implicit Integration Algorithms
  - Directly applicable for 3D/sub-space loading
  - Generalized Material Symmetry Operators (which influence flow, hardening, recovery, relaxation spectrum, etc.)
  - Efficiency (through explicit algorithmic tangent stiffness)
  - Robustness (through “slack” line search)
- Now Commercially Available
  - COMPARE
  - GVIPS – via UMATs





[www.openchannelfoundation.org](http://www.openchannelfoundation.org)



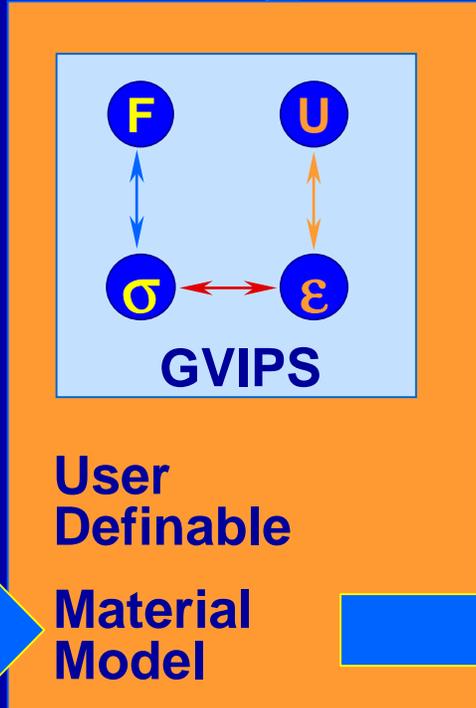
Multiple Experiments produce data

3	37446.27292	0.838069
3	37446.27292	0.838069
3	37446.27292	0.838069
3	37448.29514	1.320609
3	37450.27153	1.793447
3	37453.27222	1.005082
3	37455.28194	1.068973
3	37457.26597	1.040952
3	37460.45486	1.268637
3	37461.67014	1.357503
3	37462.34931	1.089025
3	37464.27778	1.13265
3	37467.28403	1.096359
3	37469.27153	1.064865

**Data**



COMPARE fits the GVIPS material parameters to experimental data within minutes.

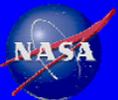


The resulting UMAT can be immediately accessed by the Finite Element Analysis



# ***Future Work***

- Extend formulation to account for
  - Coupled Nonisothermal Issues
  - Probabilistic Material Behavior
- Characterize additional material systems
- Verify under prototypical loading histories
- Implement softening (damage) mechanisms into COMPARE – theory complete
  - Characterize strength/stiffness reduction parameters to account for softening effects



# Thank You

# Questions?



# Selected References

- Arnold, S. M., and Saleeb, A.F., “On the Thermodynamic Framework of Generalized Coupled Thermoelastic-Viscoplastic -Damage Modeling”, *Jnl of Int. Plasticity*, Vol. 10, No. 3.,pp. 263-278, 1994
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- Arnold, S. M., Saleeb, A.F., Castelli, M.G., “A Fully Associative, Nonisothermal, Non-Linear Kinematic, Unified Viscoplastic Model For Titanium Based Matrices”, *Thermo-Mechanical Fatigue Behavior of Materials: Second Volume*, ASTM STP 1263, M. Verrilli and M.G. Castelli, Eds. , 1996, pp.146-173.
- Saleeb, A.F. and Arnold, S. M.; “A General Reversible Hereditary Constitutive Model: Part I Theoretical Developments”, *JEMT*, Vol. 123, 2001, pp.51-64.
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- Saleeb, A.F. and Arnold, S.M. ; “Specific Hardening Function Definition and Characterization of A Multimechanism Generalized Potential-Based Viscoelastoplasticity Model”, accepted *Int. Jnl of Plasticity*, 2003

